not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a 'significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. App. 1354(a), 1421 and 1423; 49 U.S.C. 106(g); and 14 CFR 11 89.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

95–12–10 Airbus Industrie: Amendment 39–9260. Docket 95–NM–64–AD.

Applicability: Model A330 series airplanes having manufacturer's serial number (MSN) 030, 037, and 045; and Model A340 series airplanes having MSN 005 through 031 inclusive, and 038; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must use the authority provided in paragraph (b) to request approval from the FAA. This approval may address either no action, if the current configuration eliminates the unsafe condition; or different actions necessary to address the unsafe condition described in this AD. Such a request should include an assessment of the

effect of the changed configuration on the unsafe condition addressed by this AD. In no case does the presence of any modification, alteration, or repair remove any airplane from the applicability of this AD.

Compliance: Required as indicated, unless accomplished previously. To prevent the electrical cables in the crew/passenger door from restraining the evacuation slide/raft pack and preventing proper deployment of the slide/raft, accomplish the following:

(a) Within 450 hours time-in-service after the effective date of this AD, conduct a one-time inspection to determine the correct installation of the electrical cable wiring on the right- and left-hand escape slide rafts, in accordance with Airbus All Operators Telex (AOT) 25–08, dated April 25, 1994. Prior to further flight, correct any discrepancies identified in the electrical cable wiring installation in accordance with the Airbus AOT

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Standardization Branch, ANM–113, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Standardization Branch, ANM–113.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Standardization Branch, ANM–113.

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(d) The inspection and correction of discrepancies shall be done in accordance with Airbus All Operators Telex (AOT) 25–08, dated April 25, 1994. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Airbus Industrie, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

(e) This amendment becomes effective on June 28, 1995.

Issued in Renton, Washington, on June 1, 1995.

Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 95–13887 Filed 6–12–95; 8:45 am]

14 CFR Part 39

[Docket No. 94-NM-142-AD; Amendment 39-9257; AD 95-12-07]

Airworthiness Directives; Airbus Model A340–211 and –311 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; request for comments.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain Airbus Model A340–211 and –311 airplanes. This action requires replacement of the inboard and outboard aileron servo controls associated with hydraulic systems with new units that contain an improved seal installation. This amendment is prompted by reports of external leakage of hydraulic fluid on the inboard and/or outboard aileron servo controls on in-service airplanes. The actions specified in this AD are intended to prevent loss of hydraulic fluid, which may lead to the loss of the corresponding hydraulic system and its associated functions, and reduced controllability of the airplane.

DATES: Effective on June 28, 1995.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of June 28, 1995.

Comments for inclusion in the Rules Docket must be received on or before August 14, 1995.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM–103, Attention: Rules Docket No. 94–NM–142–AD, 1601 Lind Avenue, SW., Renton, Washington 98055–4056.

The service information referenced in this AD may be obtained from Airbus Industrie, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Stephen Slotte, Aerospace Engineer, Standardization Branch, ANM–113, FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington 98055–4056; telephone (206) 227–2797; fax (206) 227–1320.

SUPPLEMENTARY INFORMATION: The Direction Générale de l'Aviation Civile (DGAC), which is the airworthiness authority for France, recently notified